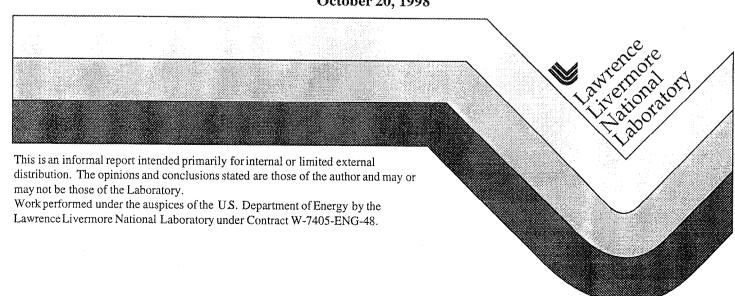
Calculating Contained Firing Facility (CFF) Explosive Firing Zones

J. W. Lyle



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CALCULATING CONTAINED FIRING FACILITY (CFF) EXPLOSIVE FIRING ZONES

J. W. Lyle *

October 20, 1998

I. INTRODUCTION

The University of California awarded LLNL contract No. B345381 for the design of the facility to Parsons Infrastructure & Technology, Inc., of Pasadena, California. The Laboratory specified that the firing chamber be able to withstand repeated firings of 60 Kg of explosive located in the center of the chamber, 4 feet above the floor, and repeated firings of 35 Kg of explosive at the same height and located anywhere within 2 feet of the edge of a region on the floor called the anvil. Other requirements were that the chamber be able to accommodate the penetrations of the existing bullnose of the Bunker 801 flash X-ray machine and the roof of the underground camera room.

These requirements and provisions for blast-resistant doors formed the essential basis for the design. The design efforts resulted in a steel-reinforced concrete structure measuring (on the inside) 55 x 51 feet by 30 feet high. The walls and ceiling are to be approximately 6 feet thick. Because the 60-Kg charge is not located in the geometric center of the volume and a 35-Kg charge could be located anywhere in a prescribed area, there will be different dynamic pressures and impulses on the various walls, floor, and ceiling, depending upon the weights and locations of the charges.

The detailed calculations and specifications to achieve the design criteria were performed by Parsons and are included in Reference 1.

Reference 2, Structures to Resist the Effects of Accidental Explosions (TM5-1300), is the primary design manual for structures of this type. It includes an analysis technique for the calculation of blast loadings within a cubicle or containment-type structure. Parsons used the TM5-1300 methods to calculate the loadings on the various firing chamber surfaces for the design criteria explosive weights and locations. At LLNL the same methods were then used to determine the firing zones for other weights and elevations that would give the same or lesser loadings. Although very laborious, a hand calculation of the different variables is possible, and an example is given in Appendix C. Fortunately, a code called "SHOCK" is available to perform these calculations rapidly, and the code runs on a personal computer. The original code was developed by the firm Amman and Whitney, which they called "Paimpres"; this was modified to its present form by the U.S. Naval Civil Engineering Laboratory. Parsons used the SHOCK code extensively, as well as several single- and multiple-degree-of-freedom codes, which were provided by the U.S. Corps of Engineers. In addition, Parsons based their analysis/design on procedures stipulated in the publication DOE/TIC-11268, A Manual for the Prediction of Blast and Fragment Loadings on Structures.

Loadings on structures in Reference 2 and in calculations performed with the SHOCK code are based on weights of explosives in pounds of TNT equivalent. The equivalency of an explosive (for its blast effects on structures) is calculated by the ratio of its heat to detonation to that of TNT. We intend to use the explosive C-4 for testing the response of the firing chamber. Various values of the ratio for C-4 are available: Reference 2 lists numbers leading to a ratio of 1.15, while 1.13 is the ratio calculated from numbers given in the LLNL Explosives Handbook (Reference 3). Parsons used a ratio value of 1.3 for generic high explosive-to-TNT equivalency. For design purposes, Reference 2 recommends a 20 percent increase in explosive weight. Parsons adopted this recommendation. Therefore, for calculational purposes, 60 Kg of generic high explosive was taken to be equivalent to 206.3 pounds of TNT. That is, 60 Kg x 2.204 lb/Kg x $1.3 \times 1.2 = 206.3$ lb (TNT).

II. CALCULATIONAL DETAILS

In section 2-14.2.1. of Reference 2, it is written:

An approximate method for the calculation of the internal shock pressures has been developed using theoretical procedures based on semi-empirical blast data and on the results of response tests on slabs. The calculated average shock pressures have been compared with those obtained from the results of tests of a scale-model steel cubicle and have shown good agreement for a wide range of cubicle configurations. This method consists of the determination of the peak pressures and impulses acting at various points of each interior surface and then integrating to obtain the total shock load. In order to simplify the calculation of the response of a protective structure wall to these applied loads, the peak pressures and total impulses are assumed to be uniformly distributed on the surface. The peak average pressure and the total average impulse are given for any wall surface. The actual distribution of the blast loads is highly irregular, because of the multiple reflections and time phasing and results in localized high shear stresses in the element. The use of the average blast loads, when designing, is predicated on the ability of the element to transfer these localized loads to regions of lower stress. Reinforced concrete with properly designed shear reinforcement and steel plates exhibit this characteristic.

The procedure for the determination of the shock loads was programmed for solutions on a digital computer. In Reference 2, the results are presented for the average peak reflected pressures on 48 figures and for the average scaled unit impulse on 48 more figures. Fortunately, the text and figures of TM5-1300 are available as computer software which includes the means to read the curves. In this way it is easy to obtain peak reflected pressure, impulse, pulse length, and other variables as a function of the scaled distance

 $(Z = \frac{R}{W^{1/3}})$. In this expression, R is the distance from charge center to the surface in question and W is

the weight of explosive in equivalent TNT pounds. The CFF firing chamber geometry is included in the range of the plotted variables so that extrapolation was not necessary, but interpolation between as many as six curves was required to fit some charge locations. The use of the SHOCK code greatly assists this process. Again from Reference 2: "The wall (if any) parallel and opposite to the surface in question has a negligible contribution to the shock loads for the range of parameters used and was therefore not considered."

The analysis leads to the conclusion that each of the chamber surfaces can be characterized by the peak pressure it would experience. For the floor, it is obvious that the maximum charge at 4 feet elevation yields the highest pressure, which at a point directly beneath the charge would be approximately 15,541 psi. The same charge gives the highest pressure on the ceiling, approximately 78 psi. The various locations of a 120.3-lb charge two feet inside the anvil edge yield the highest pressures on the walls according to the following table.

TABLE I SHOCK CODE CALCULATIONS OF PRESSURE ON THE VARIOUS SURFACES

Explosive Charges at 4-foot elevation

Surface	Pressure at Point Nearest Charge psi	Maximum Average Pressure on Entire Surface psi	Caused by
Floor	15,541.3	400.9	206.3 lb TNT, centered
Ceiling	163.5	78	206.3 lb TNT, centered
W wall	2,408.7	234.7	120.3 lb TNT, 8.53 feet from wall (as if no bullnose were present)
E wall	664.2	126.9	120.3 lb TNT, 13 feet from wall
N wall	2,602.9	209	120.3 lb TNT, at NE corner of the anvil
S wall	229.7	80	120.3 lb TNT, centered, 19.7 feet from S wall

Initially, a series of calculations was performed for the 4-foot elevation to determine where 60 Kg of explosive could be placed so that the design criteria pressures on the walls would not be exceeded. The area on the anvil for the location of 35-Kg charges was taken as given. The analysis was then extended at 5-Kg intervals to fill in the space on the anvil between 35 Kg and 60 Kg. The same analysis also located the 1-Kg line. Near the north wall there is a trench in the floor, so we thought it prudent to move the 1-Kg line a foot or so inwards to avoid bending the trench cover plates. Intermediate explosive weights between the north 1-Kg line and the 35-Kg line on the anvil were placed linearly between these two boundaries as an expedient even though it is recognized that the analysis is non-linear. In order to minimize stresses on the camera room roof, the area south of the 35-Kg line is limited to 1.5 Kg. After discussions with Larry Simmons, it was agreed that the maximum distributed explosive charge on top of the underground camera room would be 1.5 Kg. We derived this quantity by considering that, at most, 10 mirror pads would be used on a single shot. The pad's present explosive weight is 75 grams, but anticipated future design modifications may call for up to 150 grams each.

The SHOCK code features a reduced area calculation. This scheme was used to calculate the pressures and the impulse on the inner door frame of the equipment blast door as if the door were there rather than its actual location at the outer wall, 6 feet further away from the explosive. The results were then used to adjust the map profiles so that the design criteria of pressure and *impulse* at the virtual door would not be exceeded for any explosive weight or location.

The explosive weights at elevations less than 4 feet were calculated so that the pressures on the floor would not exceed 15,541.3 psi. The explosive weights at elevations greater than 4 feet were calculated so that the pressure on the ceiling would not exceed 78 psi.

III. CONCLUSIONS

The curves in Appendix A are meant to be used by ramrods, physicists, engineers, and bunker personnel for the safe placement of explosives at the facility. The objective is to maintain a minimum safety factor of 1.7 to the elastic limit for the most heavily stressed chamber element. It is important to point out that the curves are based solely on calculations. The firing chamber will be fitted with gauges to measure strain. As testing and operating experience are accumulated, it is quite likely that the map profiles will be adjusted. In addition, deviations from these maps are possible with appropriate analysis, approval, and planning and through the use of blast attenuation and mitigation measures.

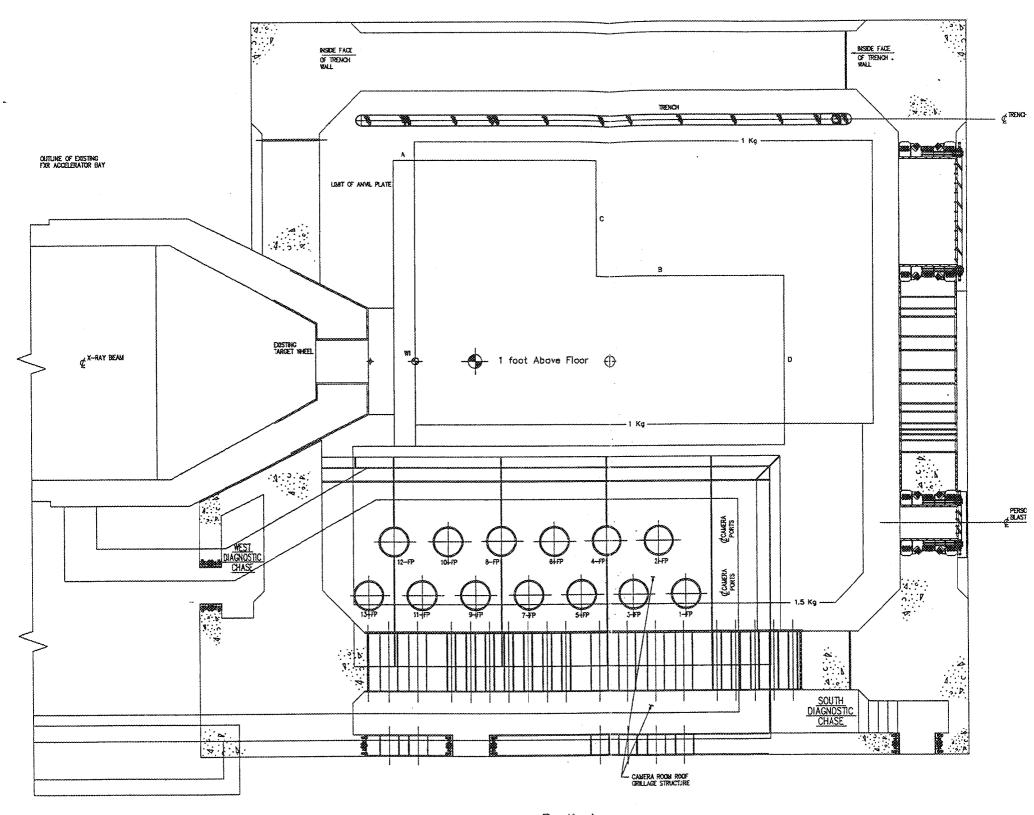
The most heavily stressed element in the firing chamber will be the floor. Various configurations of attenuating materials have been tested that minimize blast damage to the floors of explosive testing chambers. Some examples can be found in Reference 4. Additional experimental studies are now being planned as the basis for the design of an attenuating system for use in the firing chamber. Our intention is to use such a system until experience shows that it may not be necessary.

IV. REFERENCES

- 1. Site 300, Contained Firing Facility, Final Structural Chamber Calculations, LLNL Contract No. B345381, Parsons Job No. 732925, May 29, 1998.
- 2. Departments of the Army, Navy, and Air Force, Structures to Resist the Effects of Accidental Explosions, Army TM 5-1300, Headquarters, Washington, D.C., 19 November 1990.
- 3. B. M. Dobratz and P. C. Crawford, *LLNL Explosives Handbook, Properties of Chemical Explosives and Explosive Simulants*, Lawrence Livermore National Laboratory, Livermore, California, UCRL-52997 Change 2, January 31, 1985.
- 4. J. W. Pastrnak, C. F. Baker, and L. F. Simmons, *Quarter-Scale Close-in Blast-Loading Experiments in Support of the Planned Contained Firing Facility*, Lawrence Livermore National Laboratory, Livermore, California, UCRL-JC-116822, July 27, 1994.

APPENDIX A

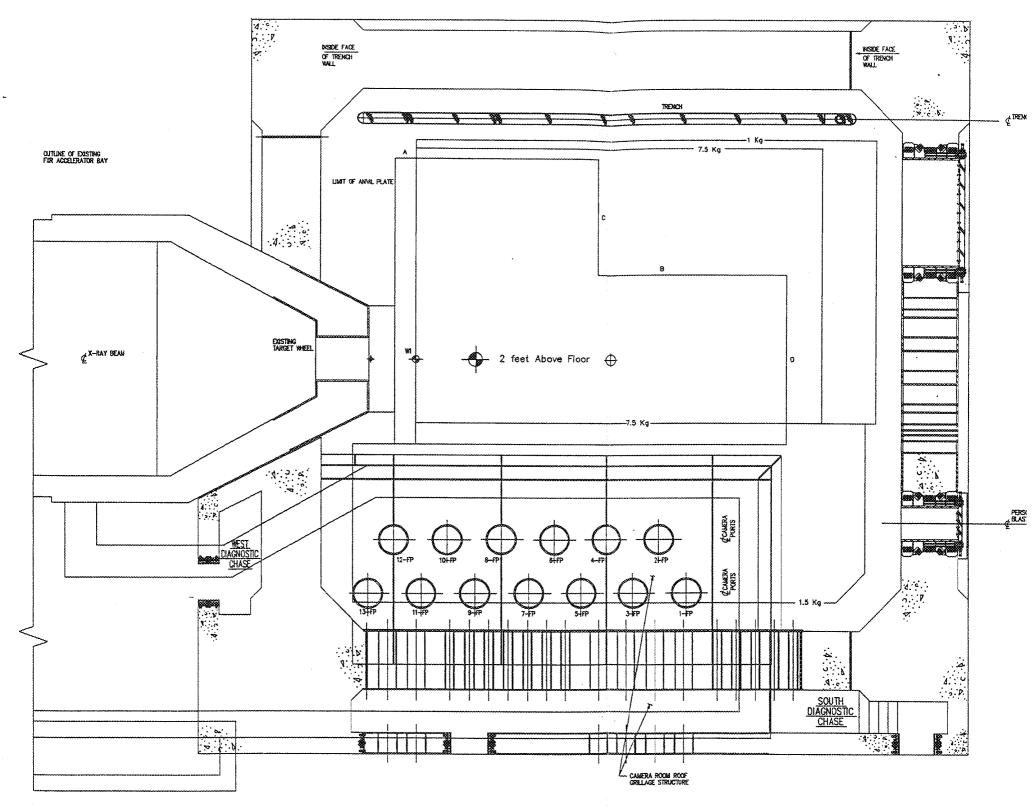
Explosive firing zone maps are given for six elevations 1, 2, 3, 3.5, 4, 8, and 12 feet above the floor. The region to the south of the 35-Kg line, over the camera room roof, is limited to a total distributed explosive weight of 1.5 Kg. This will accommodate 10 optical turning mirror explosive pads of 150 grams each.



Preliminary

CFF - HARDENED CHAMBER EXPLOSIVE FIRING ZONES

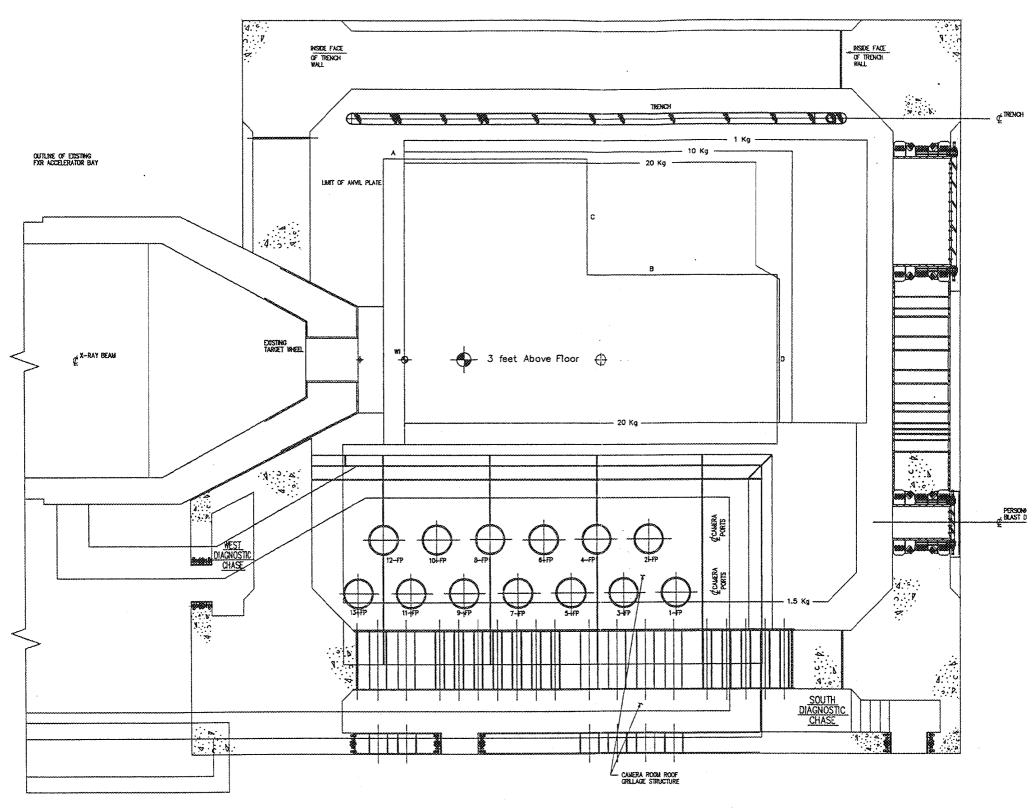




Preliminary

CFF - HARDENED CHAMBER EXPLOSIVE FIRING ZONES

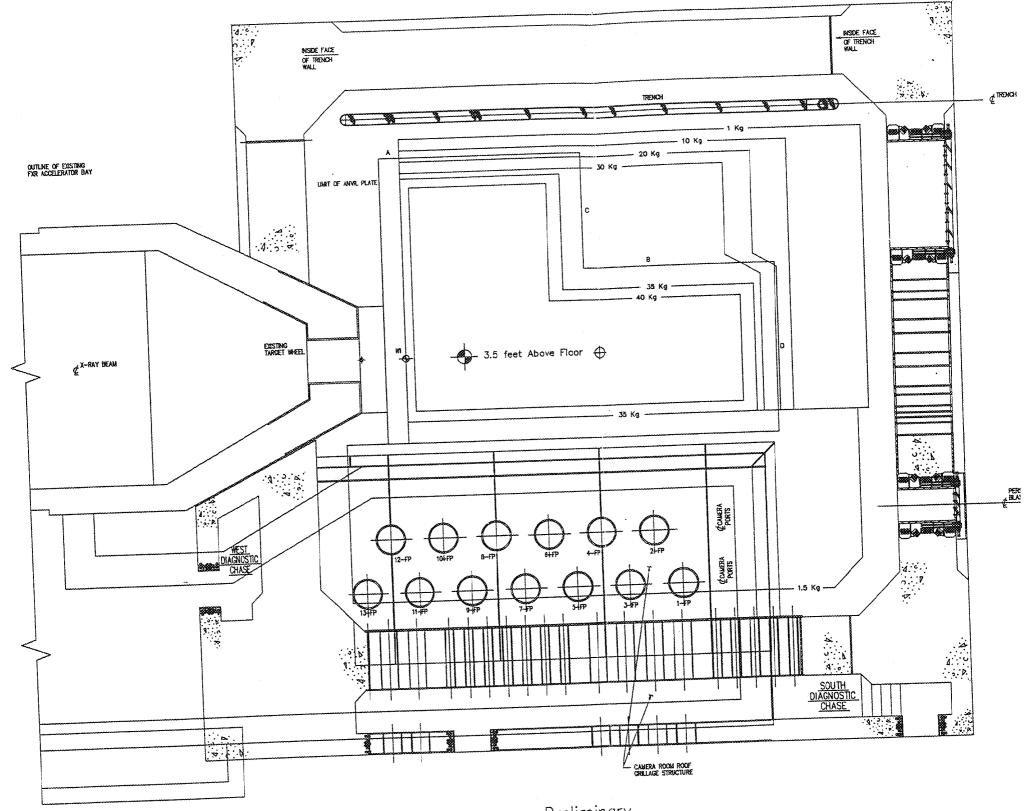




Preliminary

CFF - HARDENED CHAMBER EXPLOSIVE FIRING ZONES

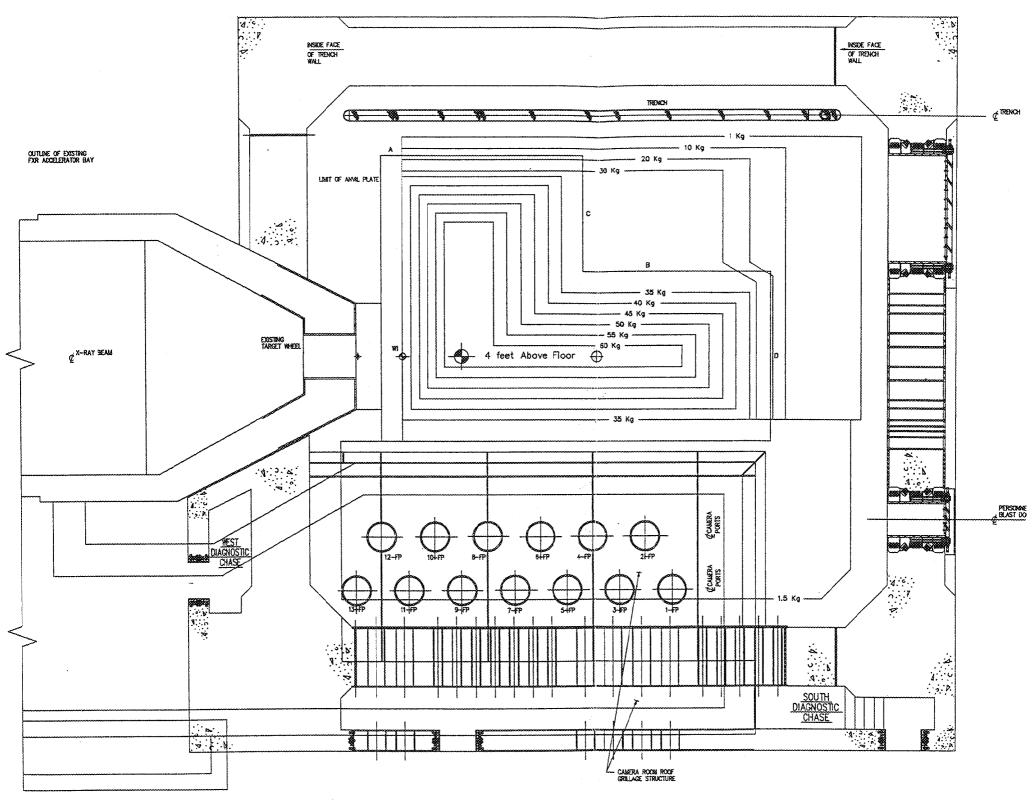




Preliminary

CFF - HARDENED CHAMBER EXPLOSIVE FIRING ZONES

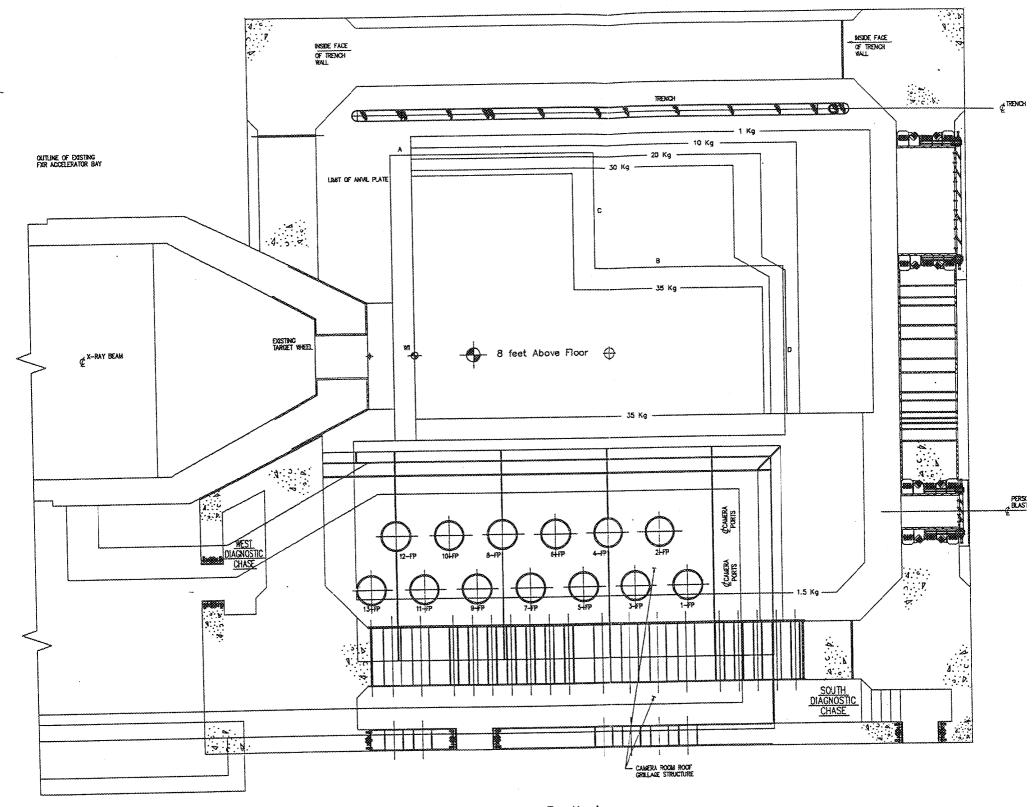




Preliminary

CFF - HARDENED CHAMBER EXPLOSIVE FIRING ZONES

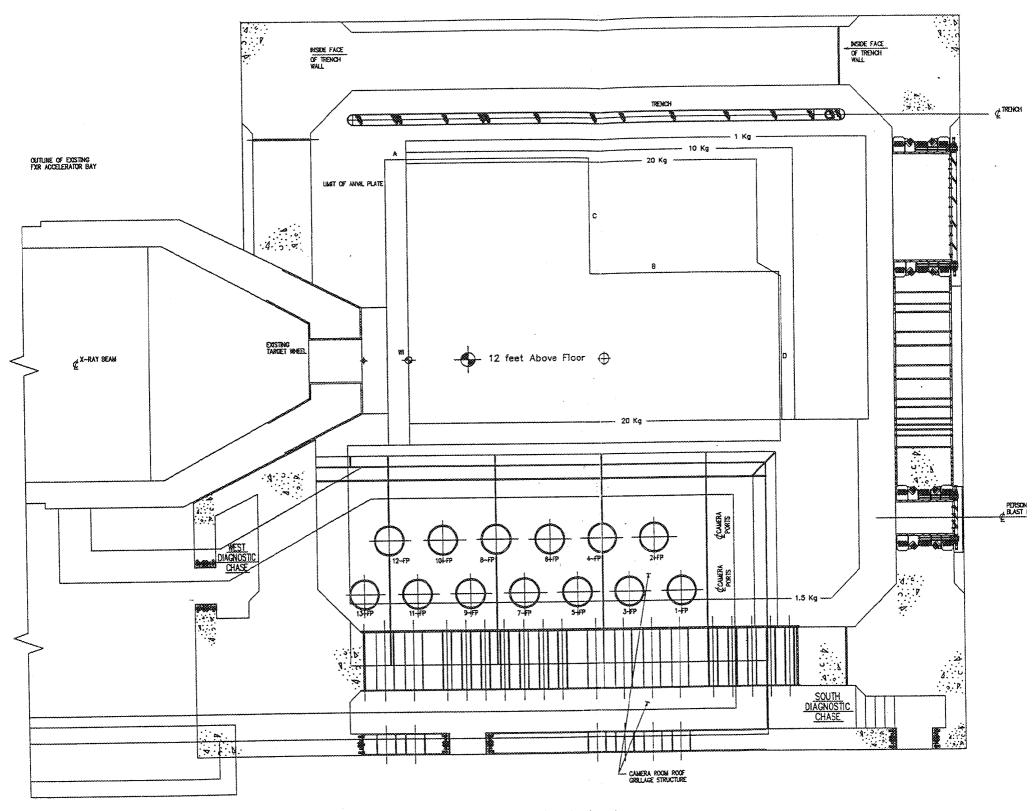




Preliminary

CFF - HARDENED CHAMBER EXPLOSIVE FIRING ZONES





Preliminary

CFF - HARDENED CHAMBER EXPLOSIVE FIRING ZONES



APPENDIX B

The SHOCK code calculations for the 206.3-lb charge of TNF are given for the floor and roof of the firing chamber to illustrate the technique and because this charge results in the highest loading on the respective surfaces. This is followed by calculations for the 120.3-lb charge, giving the maximum pressures on the east and west walls (no bullnose accounted for). One of a series of code calculations is given to illustrate the reduced area feature of the code. In this case, a virtual blast door on the inside of the chamber wall is being considered. The two remaining plots are the calculated peak average pressures and impulses on the virtual door from charges of various weights as they are moved along a bisecting normal line to the door.

PARSONS A LOCATION 206 # TNT, CENTERED

PROGRAM SHOCK VERSION 1.0

MODIFIED TO "SHOCK" BY NAVAL CIVIL ENGINEERING LAB INPUT DATA

DATA SET TITLE: A206FS FLOOR

1

Α.	CHARGE WEIGHT, LBS	206.30
В.	DISTANCE TO BLAST SURFACE, FT	4.00
С.	WIDTH OF BLAST SURFACE, FT	55.00
5	TIPTOTE OF THE CONTROLLY II	
υ.	HEIGHT OF BLAST SURFACE, FT	51.00
Ε.	HORIZONTAL (X) DISTANCE TO CHARGE	
	FROM REFLECTING SURFACE NO. 2, FT	27.50
F.	VERTICAL (Y) DISTANCE TO CHARGE	
	FROM REFLECTING SURFACE NO. 1, FT	25.50
G.	REFLECTING SURFACES	
	"1" FOR FULL REFLECTION, "0" FOR NONE	
	SURFACE 1 (FLOOR)	1
	SURFACE 2 (LEFT SIDEWALL)	1
	SURFACE 3 (CEILING)	1
	SURFACE 4 (RIGHT SIDEWALL)	1
Η.	REDUCED SURFACE CALCULATION	NO

ANALYSIS RESULTS

				~~~~	
	AVERAGE SHO	CK PRESSU	RE AND SCA	LED SHOCK	IMPULSE ON BLAST SURFACE
	DUE TO W	AVES OFF	REFLECTING	SURFACES	DUE TO INCIDENT WAVE
SURFACE	1	2	3	4	,
IMPULSE	8.6	7.9	8.6	7.9	24.3
PRESSURE	16.9	14.3	16.9	14.3	400.9

MAXIMUM AVERAGE SHOCK PRESSURE AND TOTAL AVERAGE SHOCK IMPULSE ON BLAST SURFA

SCALED IMPULSE

57.2

IMPULSE

338.2

PRESSURE

400.9

IMPULSE DURATION ON BLAST SURFACE = 1.69 MS

SCALED IMPULSES HAVE BEEN DIVIDED BY W**(1/3) = 5.91

SCALED IMPULSES ARE PSI-MS/LBS**1/3, IMPULSES ARE PSI-MS, PRESSURES ARE PSI

1

## PROGRAM SHOCK VERSION 1.0

*********************

PROGRAM FOR CALCULATION OF AVERAGE BARRIER REFLECTED SHOCK PRESSURES AND IMPULSES DUE TO AN INCIDENT WAVE AND REFLECTED WAVES FROM ONE TO FOUR REFLECTION SURFACES. ORIGINAL PROGRAM "PAIMPRES" DEVELOPED BY AMMANN AND WHITNEY MODIFIED TO "SHOCK" BY NAVAL CIVIL ENGINEERING LAB

## INPUT DATA

## DATA SET TITLE: A206RS ROOF

-		
Α.	CHARGE WEIGHT, LBS	206.30
В.	DISTANCE TO BLAST SURFACE, FT	26.00
	WIDTH OF BLAST SURFACE, FT	55.00
	HEIGHT OF BLAST SURFACE, FT	51.00
	HORIZONTAL (X) DISTANCE TO CHARGE	
	FROM REFLECTING SURFACE NO. 2, FT	27.50
F.	VERTICAL (Y) DISTANCE TO CHARGE	
	FROM REFLECTING SURFACE NO. 1, FT	25.50
G.	REFLECTING SURFACES	
	"1" FOR FULL REFLECTION, "0" FOR NONE	
	SURFACE 1 (FLOOR)	1
	SURFACE 2 (LEFT SIDEWALL)	1
	SURFACE 3 (CEILING)	1
	SURFACE 4 (RIGHT SIDEWALL)	1
н.	REDUCED SURFACE CALCULATION	NO

## ANALYSIS RESULTS

AVERAGE SHOCK PRESSURE AND SCALED SHOCK IMPULSE ON BLAST SURFACE DUE TO WAVES OFF REFLECTING SURFACES DUE TO INCIDENT WAVE SURFACE

DOMESTICE	٠,	۵	J	-T	
IMPULSE	12.7	10.8	.12.7	10.8	20.0
PRESSURE	33.6	25.3	33.6	25.3	78.0

MAXIMUM AVERAGE SHOCK PRESSURE AND TOTAL AVERAGE SHOCK IMPULSE ON BLAST SURFA

SCALED IMPULSE 67.0 IMPULSE 395.6

PRESSURE 78.0

IMPULSE DURATION ON BLAST SURFACE =10.14 MS SCALED IMPULSES HAVE BEEN DIVIDED BY W**(1/3) = 5.91SCALED IMPULSES ARE PSI-MS/LBS**1/3, IMPULSES ARE PSI-MS, PRESSURES ARE PSI

## PROGRAM SHOCK VERSION 1.0

********************

PROGRAM FOR CALCULATION OF AVERAGE BARRIER REFLECTED SHOCK PRESSURES AND IMPULSES DUE TO AN INCIDENT WAVE AND REFLECTED WAVES FROM ONE TO FOUR REFLECTION SURFACES.
ORIGINAL PROGRAM "PAIMPRES" DEVELOPED BY AMMANN AND WHITNEY MODIFIED TO "SHOCK" BY NAVAL CIVIL ENGINEERING LAB

## INPUT DATA

## DATA SET TITLE: EW35KG

1

A. CHARGE WEIGHT, LBS...... 120.12 B. DISTANCE TO BLAST SURFACE, FT..... 13.00 C. WIDTH OF BLAST SURFACE, FT..... 51.00 D. HEIGHT OF BLAST SURFACE, FT..... 30.00 E. HORIZONTAL (X) DISTANCE TO CHARGE FROM REFLECTING SURFACE NO. 2, FT...... 19.50 F. VERTICAL (Y) DISTANCE TO CHARGE FROM REFLECTING SURFACE NO. 1, FT..... 4.00 G. REFLECTING SURFACES "1" FOR FULL REFLECTION, "0" FOR NONE SURFACE 1 (FLOOR)...... 1. SURFACE 2 (LEFT SIDEWALL)..... SURFACE 3 (CEILING)...... SURFACE 4 (RIGHT SIDEWALL)..... 1 H. REDUCED SURFACE CALCULATION..... NO

## ANALYSIS RESULTS

AVERAGE SHOCK PRESSURE AND SCALED SHOCK IMPULSE ON BLAST SURFACE
DUE TO WAVES OFF REFLECTING SURFACES DUE TO INCIDENT WAVE
1 2 3 4

 SURFACE
 1
 2
 3
 4

 IMPULSE
 19.5
 9.3
 9.3
 6.6
 21.8

 PRESSURE
 105.7
 19.4
 18.1
 10.3
 126.9

MAXIMUM AVERAGE SHOCK PRESSURE AND TOTAL AVERAGE SHOCK IMPULSE ON BLAST SURFA

SCALED IMPULSE 66.5 IMPULSE 328.2

PRESSURE 126.9

IMPULSE DURATION ON BLAST SURFACE = 5.18 MS

SCALED IMPULSES HAVE BEEN DIVIDED BY W**(1/3) = 4.93

SCALED IMPULSES ARE PSI-MS/LBS**1/3, IMPULSES ARE PSI-MS, PRESSURES ARE PSI

# PARSONS B LOCATION 120,3 H TNT

## PROGRAM SHOCK VERSION 1.0

************************ PROGRAM FOR CALCULATION OF AVERAGE BARRIER REFLECTED SHOCK PRESSURES AND IMPULSES DUE TO AN INCIDENT WAVE AND REFLECTED WAVES FROM ONE TO FOUR REFLECTION SURFACES. ORIGINAL PROGRAM "PAIMPRES" DEVELOPED BY AMMANN AND WHITNEY MODIFIED TO "SHOCK" BY NAVAL CIVIL ENGINEERING LAB

### INPUT DATA

## DATA SET TITLE: B120WW

1

A.	CHARGE WEIGHT, LBS	120.30
В.	DISTANCE TO BLAST SURFACE, FT	8.53
C.	WIDTH OF BLAST SURFACE, FT	51.00
D.	HEIGHT OF BLAST SURFACE, FT	30.00
E.	HORIZONTAL (X) DISTANCE TO CHARGE	
	FROM REFLECTING SURFACE NO. 2, FT	25.50
F.	VERTICAL (Y) DISTANCE TO CHARGE	
	FROM REFLECTING SURFACE NO. 1, FT	4.00
G.	REFLECTING SURFACES	
	"1" FOR FULL REFLECTION, "0" FOR NONE	
	SURFACE 1 (FLOOR)	1
	SURFACE 2 (LEFT SIDEWALL)	1
	SURFACE 3 (CEILING)	1
	SURFACE 4 (RIGHT SIDEWALL)	1
Η.	REDUCED SURFACE CALCULATION	NO

### ANALYSIS RESULTS

AVERAGE SHOCK PRESSURE AND SCALED SHOCK IMPULSE ON BLAST SURFACE DUE TO WAVES OFF REFLECTING SURFACES DUE TO INCIDENT WAVE SURFACE 1 2 3 IMPULSE 20.1 7.4 8.9 7.4 24.3 PRESSURE 180.3 12.8 16.8 12.8 234.7

MAXIMUM AVERAGE SHOCK PRESSURE AND TOTAL AVERAGE SHOCK IMPULSE ON BLAST SURFA

SCALED IMPULSE 68.2 IMPULSE

PRESSURE

336.4

234.7

IMPULSE DURATION ON BLAST SURFACE = 2.87 MS SCALED IMPULSES HAVE BEEN DIVIDED BY W**(1/3) = 4.94SCALED IMPULSES ARE PSI-MS/LBS**1/3, IMPULSES ARE PSI-MS, PRESSURES ARE PSI

# 35 KG NEAREST EAST WALL ON ANVIL REDUCED AREA - DOOR FRAME

## PROGRAM SHOCK VERSION 1.0

******************

PROGRAM FOR CALCULATION OF AVERAGE BARRIER REFLECTED SHOCK PRESSURES AND IMPULSES DUE TO AN INCIDENT WAVE AND REFLECTED WAVES FROM ONE TO FOUR REFLECTION SURFACES.

ORIGINAL PROGRAM "PAIMPRES" DEVELOPED BY AMMANN AND WHITNEY MODIFIED TO "SHOCK" BY NAVAL CIVIL ENGINEERING LAB

#### INPUT DATA

DATA SET TITLE: EW35DR

1

Α.	CHARGE WEIGHT, LBS	120.12	
В.	DISTANCE TO BLAST SURFACE, FT	13.00	
	WIDTH OF BLAST SURFACE, FT		
	HEIGHT OF BLAST SURFACE, FT		
	HORIZONTAL (X) DISTANCE TO CHARGE		
	FROM REFLECTING SURFACE NO. 2, FT	19.50	
F.	VERTICAL (Y) DISTANCE TO CHARGE		
	FROM REFLECTING SURFACE NO. 1, FT	4.00	
G.	REFLECTING SURFACES		
	"1" FOR FULL REFLECTION, "0" FOR NONE		
	SURFACE 1 (FLOOR)	1	
	SURFACE 2 (LEFT SIDEWALL)	. 1	
	SURFACE 3 (CEILING)	1	
	SURFACE 4 (RIGHT SIDEWALL)	1	
н.	REDUCED SURFACE CALCULATION	YES	
	CORNERS OF REDUCED AREA; X, Y; FT		
	UPPER LEFT CORNER	6.52	13.00
	UPPER RIGHT CORNER		
	LOWER LEFT CORNER		
	LOWER RIGHT CORNER		

#### ANALYSIS RESULTS

AVERAGE SHOCK PRESSURE AND SCALED SHOCK IMPULSE ON REDUCED SURFACE DUE TO WAVES OFF REFLECTING SURFACES DUE TO INCIDENT WAVE 2 SURFACE 1 3 IMPULSE 32.0 7.3 11.9 4.8 35.8 PRESSURE 266.3 25.7 11.2 5.4 303.4

MAXIMUM AVERAGE SHOCK PRES. AND TOTAL AVERAGE SHOCK IMPULSE ON REDUCED SURFAC SCALED IMPULSE 91.7

SCALED IMPULSE IMPULSE

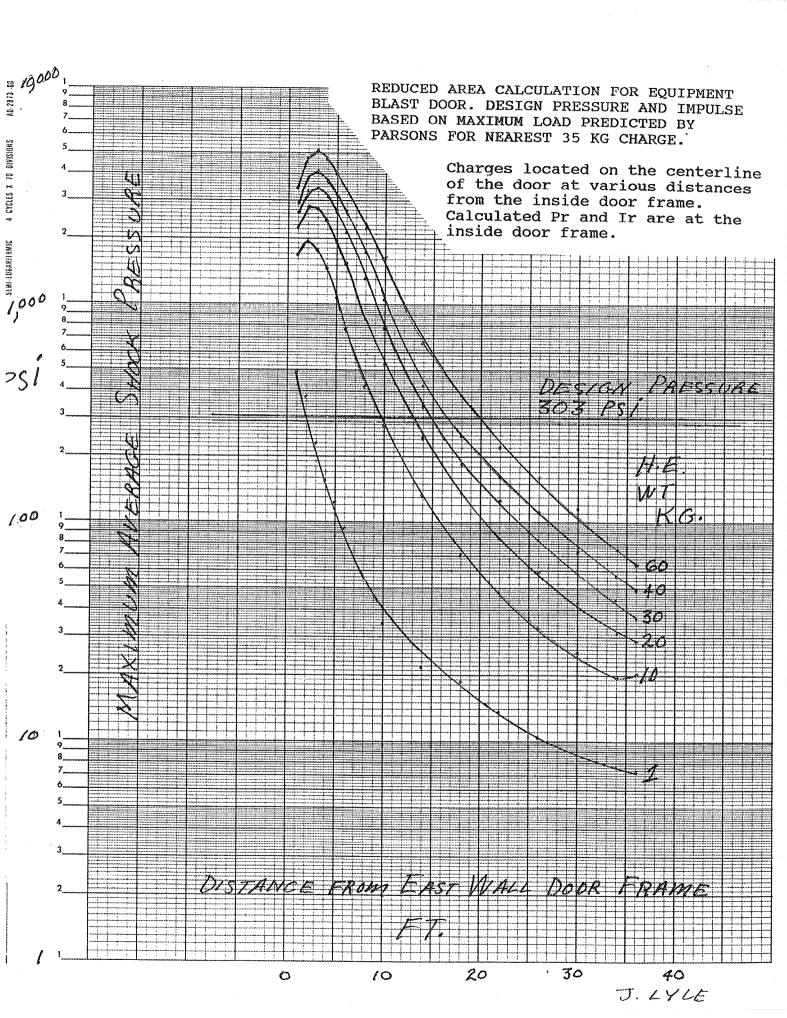
452.5

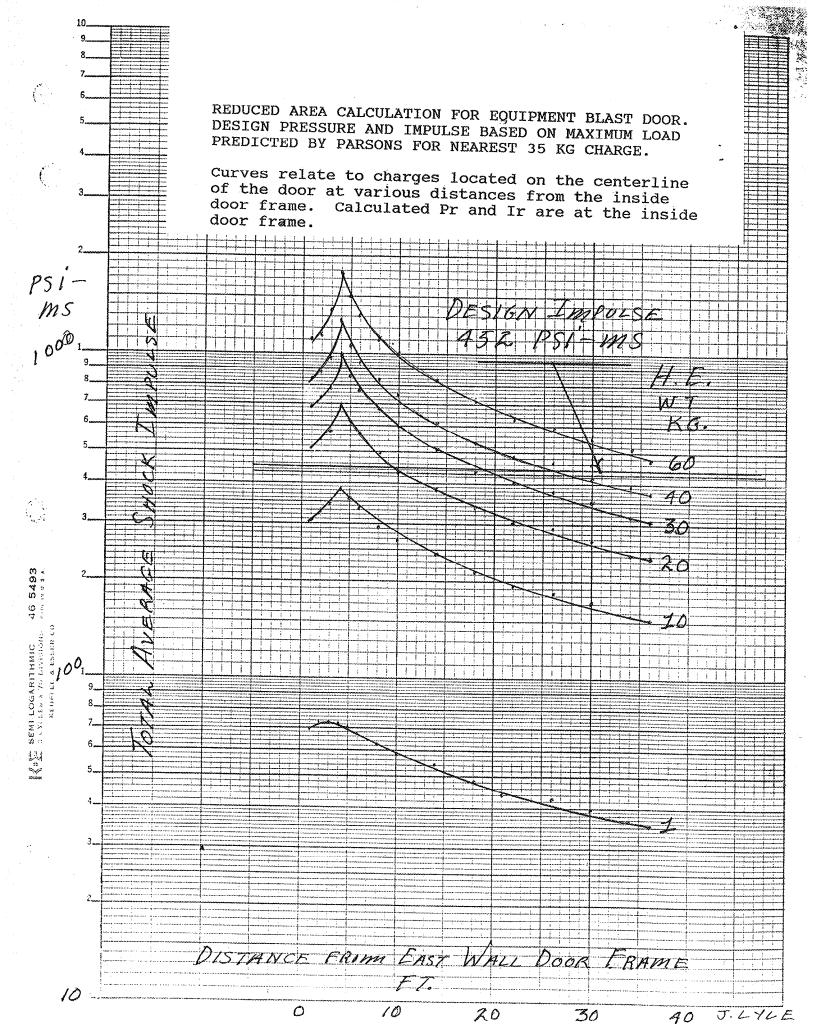
PRESSURE 303.4

IMPULSE DURATION ON BLAST SURFACE = 2.98 MS

SCALED IMPULSES HAVE BEEN DIVIDED BY W**(1/3) = 4.93

SCALED IMPULSES ARE PSI-MS/LBS**1/3, IMPULSES ARE PSI-MS, PRESSURES ARE PSI





## APPENDIX C

These hand calculations illustrate the methods of Reference 2 that can be used to calculate explosive firing zones. The use of the SHOCK code has replaced these methods principally because of its speed and its , reduced area feature which allows a determination of average shock and impulse on specified areas and points.

Fle 24.3 D

The HE. Loading Gurves
for the Firing Chamber
of the Contained Firing
Facility (CFF), Site 300,
LLNL.

C.Y. KING April 1998

ENGINEERING CALCULATION		Poga No.
	Noma	
	Data	

# contants

Topics	Page No
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II Calculations	4
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3. Vertical loadings (from 1' to 4)	11
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8 To find the distance from the floor	
where the strass of the Cailing is 4 ksi	21
9 Limiting Load above the Camera room	21
Fig. 1 Horizontal Curves & above floor	2.5
" Z " " " " " " " " " " " " " " " " " "	23
$\frac{1}{2}$ $\frac{1}{3}$ $\frac{1}{6}$ $\frac{1}{6}$ $\frac{1}{6}$ $\frac{1}{6}$ $\frac{1}{6}$ $\frac{1}{6}$ $\frac{1}{6}$ $\frac{1}{6}$ $\frac{1}{6}$	2 q.
$n = \frac{1}{4}$	25

ENGINEERING CALCULATION		Poga No.
	Nama	
•	Data	

# Summary & Conclusions

- 1. Honritontal loading Curves for 1', 2', 3' and 4' above the floor are plotted.
- 2. From 4' to 18', the horizontal curves are the same as a'.
- 3. The stand-off distance is 1-6"
- 4. Limiting load above the carema room is 150 mg to prevent damage of the Camera lens.

ENGINEERING CALCULATION		Poga No.
The H.E. Loading Curves for CFF "5,40 300.	Noma	CYKING
LLNC.	Dota	4/9/98

## I. Critaria.

- 1. At 4 above the floor.
  - a. 35 KE H.E at the adges where is 2 inside of the boundries of the anvil plates.
  - b. 60 KG H.E at the center of the chamber.
- The Maximum Concite Compressive Strass of the
  floor is 12 ksi Aksi concrete is at elsa-where.

  (Ret: Parsons Intrastructura & Tachnology Group,

  "CFF" Title I, 100% Design Review, "CFF Anvil

  and Floor Blast Analysis For "CFF", CCNC, datad

  3128/97 Appendix 4, Summory of Maximum

  Compressive Strass along the Floor Depth.")

# II. Procedures & Objections

- 1. To calculate the required horizontal distances from the inside serface of the wall, for 35 kg to 60 kg at 5 kg increment, and from 35 kg down to the stand-off distance.

  2. To determine the amount of H.E. in kg at the different elevations for the loading Curves.
- 3. To determine the stone of distance from the
- 4 The end product is esset of the H.E. loading Curves for the Firing chamber of the Contained Firing Facility (CFF) at different elevations.

ENGINEERING CALCULATION		Poge No.
	1	C.Y. KING
	Data	4/20/98

## · I Calculations

Convarsion from HE (kg) to TNT (16)

W#TNT = WK9 HE x 2.2046 x 1.3 x 1.2

Where 2.2046 is the conversion factor from Kg to Pounds.
1.3 is the factor from H.E. to TNT.

1.2 is the factor of safety (Ref. 2, Chapter 1, Article 1-5)

... W TRIT = W Kg HE X 3. 439

W Kg HE W/3 (16/2) WIGTNI 0,25 0.860 0,951 1.720 1.198 0.5 3. 439 1.509 2 6.878 1.902 2.177 3 10,317 4 2.396 13.756 5 2,581 17.195 6 20,634 2,743 7 24.073 2.887 8 3.019 27.512 9 3.140 30.951 10 34.390 3 252 15 51.585 3.723 4.097 20 68,780 25 85,975 4414 30 103,170 4.690 35 120,365 4.937

~ 1 ~ 1 A 1 ~ pm p	7101-			NATAA PARAMETER KANDA PARAMETE	Pagin Ma
ENGINEER	ING C	ALCUL	ATION		Poga No.
			A STATE OF THE STA	Noma	C.Y.KING
				Data	4/20/98
. W K9,	46	WIGTNT	W 1/3		,
40		137,560	5.162		
45		154.755	5.369		
50 '		171.950	5.561	,	
55		189.145	5.740		
60		206.740	5.909		
1. At	4 above to	ha floor,	to determi	ina the	Critical
Case					
North W	all (lafi	t partia			
			= 1.72 ft/16		/ }
				3 (Contro	
60 ng	R = 25.5'	$Z = \frac{25.3}{5.909}$	= 4,32		
	all (Righ	***************************************			
35 Kg	R=25.5-6	= 19.5	Z= 14.5 4.937 = 3	3.95 (cor	ntrol)
60 Kg	R=25,5'		$\frac{2}{5,909} = 4$	1.57	
West	Wall		5,409	102	
35 Kg	R= 9.37	= 4.37	=1.90 (CO	ntrel)	
60 Kq		7 = 27.5 7 = 27.50			
South	vall	5.70	7		
35 Kg	R=19.5	2 : 42;	5 = 3,95 (Ca	ontrol)	
	R = 25.5			,	
East i	46 M	(T. 9€	9 - 9.3 -		
<u> </u>		<u> </u>	; = 2,63 ( c	Construct 1	
60 K9			•	-UNTROID	
<i>60</i> -	14-2/13	7 5 90	= 4.65		

ENGINEERING CALCULATION		Paga No
·		- 6
	Noma Data	C.Y KIN
2. To find the required distances -		4/20/98
	TREM THE	17751de
Surface of the wall.		
North Wall (left portion)	*	
From 35 kg up		
$R = 8.5 (8-6") 2 = \frac{\epsilon.5}{4.937} = 1.72$		
R40 = 1.72 x 5.162 = 8.879' (8'-105/8")		I dimansio.
$R45 = 1.72 \times 5.369 = 9.235' (9'-27'8'')$		ned inch a
RSO= 1.72 x 5.561 = 9.565 (9-678")	in neares	f eighth.
RSS = 1.72 x 5.740 = 9.873' (9'-10/2)		
R60 = 1.72 x 5.909 = 10,163' (10'-2")		
From 35 kg down		
R30 = 1,72 x 4.690 = 8.067 (8-0/9")		
R25 = 1.72 x 4.414 = 7.592 (7-7/4")		
RLO = 1.72 x 4.097 = 7.211 (7'-2%)		
RIS = 1.72 x 3.723 = 6.404 (6-478")		
RIO = 1.72 x 3.752 = 5.593 (5'- 7/8)	•	
Rq = 1.72 x 3.140 = 5.401 (5'- 4 1/8")		
Re= 1.72 x 3,019 = 5,193 (5-23/8)		
R7 = 1.72 x 2.887 = 4.966 (4'-115/8")		
R6= 1.72x2.743=4.718 (4'-85/6")		
RS = 1.72 X2,581 = 4.439 (4'. 53/6)		
$R_4 = 1.72 \times 2.396 = 4.121 \left(4'-1/2''\right)$		
R3= 1.72 x 2.177= 3.744 (3'- 9")		
Rz= 1.72 x 1.902 = 3.271 (3'-3/4')		
R. = 1.72 x 1.509 = 2.596 (2'-7/2)		

") Nota: All dimansions in fraction of inch are in nearest eighth.

Poga No.

C.Y KING

4/20198

ENGINEERING	CALCULATION		Poga No.
		Noma	C.Y. KING
		Data	4./20/98

 $Ros = 1.72 \times 1.198 = 2.061'(2-034'')$  $Ros = 1.72 \times 0.951 = 1.636'(1-134'')$ 

North wall (Right Portion) Frem 35kg up R=19.5' == 19.5 = 3.95 R35 = 3.95 X4.937 = 19.5 (79-6") Rec = 3.95 x5.162 = 20.390 (20-43/4") R45 = 3.95 x 5.369 = 21,208 (20'-21/2") Rso = 3.95 x 5.561 = 21.966 (21-115/8) Rss = 3.95 x 5.740 = 22.673 (22-8/8) REC = 3.95 x 5.909 = 23.341 (23-4/6") From 35 kg down R30 = 3.95 K 9.690 = 18.526 (18 = 68 ) R25 = 395 x 4414 = 17.435 (17-5%) Rzo = 3.95 x 4.097 = 16.183 (16-2/11) RIS = 3.95 x 3,723 = 14,706 (14 2/4") RIO = 3.95 K 3.252 = 12.845 (12-10/2") R= 3.95 x 3.140 = 12.403 (18-0/4) Rg = 3.95 x 3.019 = 11.925 (11 - 11/2) R7 = 3.95 x 2.887 = 11.404 (11-474) R6 = 3.95 K 2.743 = 10.835 1 10-10/8) R5 = 3.95 x 2,58/= 10,195 (10 = 23/8) Ra=3.95 x 2.396 = 9.464 (9- 5%)

 $R_3 = 3.95 \times 2.177 = 8.590 (8-7%)$ 

ENGINEERING	CALCULATION		Poga No.	
		Noma	C.Y.KING	
		Data	4/20/98	

 $R_2 = 3.95 \times 1.902 = 7.513' (7-64')$   $R_1 = 3.95 \times 1.509 = 5.961' (5-11%')$   $R_{0.5} = 3.95 \times 1.198 = 4.732' (4-8%)$   $R_{0.25} = 3.95 \times 0.951 = 3.757' (3-9%)$ 

# West Wall From 35 Kg up

 $R_{35} = 1.9 \times 4.937 = 9.37' (9'-4/2)$   $R_{40} = 1.9 \times 5.162 = 9.808' (9'-9/4)$   $R_{45} = 1.9 \times 5.369 = 10.201' (10'-2/2)$   $R_{50} = 1.9 \times 5.561 = 10.566' (10'-6/2)$   $R_{55} = 1.9 \times 5.790 = 10.906' (10'-10/2'')$   $R_{10} = 1.9 \times 5.909 = 11.227' (11'-2/34'')$   $From 35-49 \ down$ 

From 35-89 down

R3c = 1.9 x 4.690 = 8.911 (8'-11")

R2s = 1.9 x 4.914 = 8.387' (8'-434")

R2c = 19 x 4.097 = 7.784' (7'-9/2")

R1s = 1.9 x 3.723 = 7.079' (7'-1")

R10 = 1.9 x 3.252 = 6.179' (6'-2/4)

R9 = 1.9 x 3.140 = 5.966' (5'-182")

Re = 1.9 x 3.019 = 5.736' (5'-872")

R7 = 1.9 x 2.887 = 5.485' (5'-572")

R6 = 1.9 x 2.743 = 5.212' (5'-252")

R5 = 1.9 x 2.581 = 4.904' (4'-1078")

R4 = 1.9x 2.396 = 4.552 (4'-65/E")

ENGINEERING CALCULATION		Poga No.
	Noma	C, Y, KING
	Data	4/20/98

R3=1.9 x 2.177 = 4.136 (4-134") Rz=1.9x1.902=3.614 (3'-73/3") R,=1.9 x1,509=2.867(2-10/3") R.s= 1.9 x 1.198 = 2.276 (2'-33/8") R. 25 = 1.9 x, 951 = 1.807 (1-9%) South Wall From 35 Kg Up R35 = 3.95 x 4.937 = 19.5 (1966")  $R40 = 3.95 \times 5.162 = 20.390' (20' - 43/4'')$   $R45 = 3.95 \times 5.369 = 21.208 (21' - 21/2')$   $R50 = 3.95 \times 5.567 = 21.966' (21 - 11/8')$ Rss = 3.95 x s. 740 = 22.673 (22'-8/8") RED = 3.95 x 5.909 = 23.341'(23'-4/6") From 35 kg down R30 = 3,95 x 4,690 = 18,526 (11 - 636") R25 = 3.95 x 4.414 = 17.435 (17-54") R20 = 3.95 x 4.097 = 16.185 (16 - 24") R15 = 3.95 x 3, 723 = 14,706 (14 - 8/3") R 10 = 3.95 x 3,252 = 12,895 (12'-10/2") Rq = 3,95 x 3,140 = 12,403 (12 - 43/2") Re = 3.95 x 3.019= 11. 925 (11 - 11/2") RT = 3,95 x Z, 887= 11,404 (1/2 4/8) RG = 3.95 x 2.743 = 10.835 (10 - 10/8") RS = 3.95 x 2.581 = 10.195 (16 - 23/4") R4 = 3.85 x 2,386 = 9.464' (91 + 5/4")

R3 = 3.95 x 2.177 = 8,599 (2 - 7/4")

ENGINEERING CALCULATION		Poga No.	
	Noma	C.Y. KING	
	Data	4/20/98	

 $R_{2} = 3.95 \times 1.902 = 7.513' (7 - 6/4'')$   $R_{1} = 3.95 \times 1.509 = 5.961' (5' - 1176'')$   $R_{.5} = 3.95 \times 1.198 = 4.732' (4' - 8/6'')$   $R_{.25} = 3.95 \times 0.951 = 3.757' (3 - 9/6'')$ 

## East Wall

From 35 kg up K35 = 2.63 x 4.937 = 13' R40 = 2.63 x 5,162 = 13 576 (13-7") Ras = 2.63 x 5.369 = 19.120 (14-1/2") R50 = 2.63 x 5.561 = 14.625 (14'-7/5") Rss = 2.63 x 5.740 = 15.096 115 - 1/2") Ren = 2,63x 5,909 = 15,501 (15'-61/6") From 35 kg down R30 = 2.63 × 4.690 = 12.335 (12-4/9) R25 = 2.63 x4,414 = 11.609 (11-73/4) Rzo = 2,63 x 4.097 = 10,775 (10'-93/2) R15 = 2.63 x 5.723 = 9.792 (9'-91/2) Rio = 2.63 x 3.252 = 8,553 (8'-63/4) R9 = 2.63 x 3,140 = 8,258 (8-5/6)  $R_8 = 2.63 \times 3.019 = 7.940/(7'-11\frac{3}{8}'')$  $R_7 = 2.63 \times 2.887 = 7.593'(7'-7/8')$ R6 = 2.63 x 2.743 = 7.214 (7'-2%)  $R_5 = 2.63 \times 2.581 = 6.788' (6'-9')$ 

R4 = 2,63 X2,396 = 6,302 (6-35/8)

ENGINEERING	CALCULATION		Poga No.	
	·		C.Y. KING	
		Data	4/21/98	

$$R_{3} = 2.63 \times 2.177. = 5.726 \quad (5-8\frac{3}{4}")$$

$$R_{2} = 2.63 \times 1.902 = 5.002 \quad (5-0\frac{3}{8}")$$

$$R_{1} = 2.63 \times 1.509 = 3.969 \quad (3'-11\frac{5}{8}")$$

$$R_{0.5} = 2.63 \times 1.198 = 3.151 \quad (3'-1\frac{7}{8}")$$

$$R_{0.5} = 2.63 \times 1.198 = 3.151 \quad (3'-1\frac{7}{8}")$$

$$R_{0.25} = 2.63 \times 1.951 = 2.501 \quad (2'-6\frac{7}{8}")$$

# 3. To determine the maximum loading at 3', 2' and 1' from the floor at 4' $35^{Kg} \quad 2 = \frac{4}{4.937} = .8102$ $60^{Kg} \quad 2 = \frac{4}{5.909} = 0.6769 \quad (Control)$ at 3' $2 = \frac{3}{N/5} = 0.6769 \quad N/5 = \frac{3}{.6769} = 4.4320$ $W = 87.0561 = 25.3144 \quad K9$ at 2' $2 = \frac{2}{N/5} = .6769 \quad N/5 = \frac{2}{.6769} = 2.9546$ $W = 25.7927 \quad ETNT = 7.5 \quad K9$ at 1' $2 = \frac{1}{N/5} = .6769 \quad N/5 = \frac{1}{.6769} = 1.4773$ $W = 3.224 \quad TNT = 0.9375 \quad K9$

4. At 3' above the floor, to find the required distances from 2' inside of anvil plate to the inside surface of the wall.

North wall (Left portion)
$$R=8.5' \quad W=4.4320 \quad Z=\frac{8.5}{4.4320}=1.9179 \text{ W/s}$$

$$25 \text{ Kg} \quad R_{25}=1.9179 \text{ Ka. 414}=8.466' (8'-55'')$$

 CALCULATION		12	No.
	Nome	~ V	10.00

Data 4/21/98

$$20 \text{ kg}$$
  $R_{20} = 1.9179 \times 4.097 = 7.8576' (7-10\%)$ 
 $15^{49}$   $R_{15} = 1.9179 \times 3.723 = 7.1403' (7-1\%)'$ 
 $10^{49}$   $R_{10} = 1.9179 \times 3.252 = 6.2370' (6'-2\%')$ 
 $5^{49}$   $R_{5} = 1.9179 \times 2.581 = 4.9560' (4'-11\%'')$ 
 $4^{49}$   $R_{4} = 1.9179 \times 2.396 = 4.5953' (4'-7\%'')$ 
 $3^{45}$   $R_{3} = 1.9179 \times 2.177 = 4.1753' (4'-2\%'')$ 
 $2^{45}$   $R_{2} = 1.9179 \times 1.902 = 3.6478' (3'-7%'')$ 
 $1^{49}$   $R_{1} = 1.9179 \times 1.509 = 2.8941' (2'-10\%'')$ 
 $0.5^{49}$   $R._{5} = 1.9179 \times 1.198 = 2.2976' (2'-3\%'')$ 
 $0.25^{45}$   $R._{25} = 1.9179 \times 1.951 = 1.8239' (1'-10")$ 

North Wall (Right portion)  $R = 19.5' \quad \text{W/3} = 4.4320 \quad 2 = \frac{19.5}{4.4320} = 4.3998$   $25 \text{ Kg} \quad R_{25} = 4.3998 \times 4.414 = 19.4207' (19'-5/8')$   $R_{20} = 4.3998 \times 4.097 = 18.0260' (18'-038')$   $R_{15} = 9.3998 \times 3.723 = 16.3805' (16'-438'')$   $R_{10} = 4.3998 \times 3.252 = 14.3083' (14'-334'')$   $R_{5} = 4.3998 \times 2.581 = 11.3559 (11'-438')$   $R_{4} = 4.3998 \times 2.394 = 10.5419 (10'-68')$   $R_{5} = 4.3998 \times 2.177 = 9.5784 (9'-7'')$   $R_{1} = 4.3998 \times 1.902 = 8.3684 (8'-41/2'')$   $R_{1} = 4.3998 \times 1.509 = 6.4393 (6'-734'')$   $R_{1} = 4.3998 \times 1.1998 = 5.2710 (5'-338'')$   $R_{1} = 4.3998 \times 1.1998 = 5.2710 (5'-338'')$   $R_{1} = 4.3998 \times 1.1998 = 5.2710 (5'-338'')$ 

#### ENGINEERING CALCULATION

Poga No. 13

Noma Data

C.Y.KING 4/21/98

West Wall

R= 9.37' W15 = 4.4320 2 = 9.37 = 2.1142

R25 = 2.1142 x 4.414 = 9.3321 (9/-4")

Rzo= 2.1142 x 4.097 = 8.6619 (8-8")

R15 = 2.1142 x 3.723 = 7.7812 (7-93/8)

Ric = 2.1142 x 3.252= 6.8754 (6-105/6)

Rs = 2.1142 x 2.581= 5.4568 (5-5/2")

R4 = 2.1142 x 2.396= 5.0656 (5- 0/2")

R) = 2.1142 x 2.177 = 4.6026 (4'-7/4")

R: = 2.1142 x 1.902 = 4.0212 (4'- 036")

RI = 2.1142 x 1.509 = 3.1903 (5'-236")

R. F = 2.1142 X 1.198 = 2.5328 (2'-61/2")

Rize = 2.1142 x . 95/= 2.0106 (2'-0/4")

South wall

R = 19.5' W'S = 4.4326  $Z = \frac{19.5'}{4.4320} = 4.3998$ 

R25 = 4.3998 x 4.414 = 19.4207 (19-5/2")

Rzo = 4.3998 x 4.097 = 18.0260' (18'-076")

RIS = 4.3998 X3.723 = 16,3805 (16-43/2")

Rio = 4.3998 x 3.252 = 14.308 ( (14 = 3)4")

Rs = 4.3498 + 2.58/ = 11 1519 (11-1/6")

 $R_{0} = 4.3998 \times 2.396 = 16.5419 \quad (10-65/8)$   $R_{1} = 4.3998 \times 1.509 = 2.6393(6-7/2)$   $R_{2} = 4.3998 \times 1.196 = 5.2710(6.39/2)$   $Ext = 4.3998 \times 4.028 = 17.7214 \quad (17-8/4)$   $R_{19} = 4.3998 \times 4.028 = 17.7214 \quad (17-8/4)$ 

Ris = 4.3998 x 3.956 = 17.4056 (17-478)

K17 = 4,3998 x3,881 = 17,0756 (17'-1") - Loading Carro Stop

Fix = 4,3998 x 3,804 = 12,7368 (16-8/8) here

From Wall to edge of anxil
4735 = 18692 =186,92

=15.5347 (15-63/1)

From edge of anvil

to edge of Grillage

= 305 " = 12"=1"

15-63/2 + 1=16-6/8

16-0% +(1-0")

=17/03/8

ENGINEERING CALCULATION		Poga No. 14
	Noma	C.Y. KING
	Data	4/21/98

East Wall R = 13'  $Z = \frac{13}{4.9310} = 2.9332$   $R_{25} = 2.9332 \times 4.414 = 12.9471 (12'-11\frac{3}{8})$   $R_{20} = 2.9332 \times 4.697 = 12.0173 (12'-0\frac{4})$   $R_{15} = 2.9332 \times 3.723 = 10.9263 (10'-11\frac{6}{8})$   $R_{10} = 2.9332 \times 5.252 = 9.5388 (9'-6\frac{6}{2}')$   $R_{10} = 2.9332 \times 2.581 = 7.5766 (7'-6\frac{7}{6})$   $R_{10} = 2.9332 \times 2.396 = 7.6279 (7'-6\frac{7}{6})$   $R_{10} = 2.9332 \times 2.177 = 6.3856 (6'-4\frac{3}{4})$   $R_{10} = 2.9332 \times 1.902 = 5.5789 (5'-7")$   $R_{10} = 2.9332 \times 1.902 = 5.5789 (5'-7")$   $R_{10} = 2.9332 \times 1.509 = 4.4262 (4'-5\frac{6}{8}')$   $R_{10} = 2.9332 \times 1.908 = 3.5140 (3'-6\frac{6}{4}')$ 

 $f_{125} = 2.9332 \times .951 = 2.7895 (2'-9/3')$ 

At 2 above the fleer, to find the required distances from 2 inside of the anvil plate to the inside surface of the wall.

North wall (Loft Portion) R = 8.5'  $W'_3 = 2.9546$   $2 = \frac{E.S}{2.9566} = 2.8769$   $K_7 = 2.8769 \times 2.387 = 8.3056$   $(8'-3)_4'')$   $R_6 = 2.8769 \times 2.703 = 7.9813$  (7'-11/6'')  $R_5 = 2.8769 \times 2.581 = 7.4253$  (7'-5/6'')  $R_6 = 2.8769 \times 2.396 = 6.8931$   $(6'-10)_4'')$   $R_3 = 2.8769 \times 2.177 = 6.2630$  (6'-3/4'')  $R_2 = 2.8769 \times 1.902 = 5.4719$   $(5'-5)_4''')$ 

ENGINEERING CALCULATION		Poga No. 15
		C.Y.KING
	Data	4/21/98

 $R_{1} = 2.8769 \times 1.509 = 4.3412 \quad (4'-4/e')$   $R_{.5} = 2.8769 \times 1.198 = 3.4465 \quad (3'.5%)$   $R_{.25} = 2.8769 \times .951 = 2.7359 \quad (2'-8%)$ 

North Wall (Right portion)  $R = 19.5' \quad \text{W} = 2.9546 \quad Z = \frac{19.5}{2.9546} = 6.5999$   $R_7 = 6.5999 \times 2.887 = 19.0539 \quad (19'-03'')$   $R_6 = 6.5999 \times 2.743 = 18.1035 \quad (18'-1/4'')$   $R_5 = 6.5999 \times 2.581 = 17.0343 \quad (17'-0/2'')$   $R_4 = 6.5999 \times 2.396 = 15.8134 \quad (15'-97'8')$   $R_5 = 6.5999 \times 2.177 = 14.3680 \quad (14'-4/2'')$   $R_1 = 6.5999 \times 1.902 = 12.5530 \quad (12'-63'')$   $R_1 = 6.5999 \times 1.509 = 9.9592 \quad (9'-115'')$   $R_5 = 6.5999 \times 1.198 = 7.9067 \quad (7'-11'')$   $R_{.25} = 6.5999 \times .951 = 6.2765 \quad (6'-33'8'')$ 

West Wall

 $R = 9.37' \quad W''^{3} = 2.9546 \qquad 2 = \frac{9.37}{2.9546} = 3.1713$   $R_{7} = 3.1713 \times 2.867 = 9.1555 \quad (9' - 176'')$   $R_{6} = 3.1713 \times 2.743 = 8.6989 \quad (8' - 8/2'')$   $R_{5} = 3.1713 \times 2.581 = 8.1851 \quad (8' - 2/4'')$   $R_{4} = 3.1713 \times 2.396 = 7.5984 \quad (7' - 7/4'')$   $R_{3} = 3.1713 \times 2.177 = 6.9639 \quad (6' - 1078'')$   $R_{4} = 3.1713 \times 1.902 = 6.4122 \quad (6' - 5'')$   $R_{5} = 3.1713 \times 1.902 = 5.0673 \quad (5' - 1/8'')$ 

ENGINEERING	CALCULATION	Poga No.	
		C.Y. KING	
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 $R_{0.5} = 3.1713 \times 1.198 = 3.7992 (3'-9%)$  $R_{0.25} = 3.1713 \times .951 = 3.6159 (3'-04")$ 

South wall R=19.5' N'5=2.9546  $Z=\frac{19.5}{2.9546}=6.5999$   $R_7=6.5999 \times 2.887=19.0539 (19-034')$   $R_6=6.5999 \times 2.743=18.1035 (18-1/4'')$   $R_5=6.5999 \times 2.581=17.0345 (17-0/2'')$   $R_4=6.5999 \times 2.396=15.8134 (15-978')$   $R_3=6.5999 \times 2.177=14.3680 (14-4/2'')$   $R_4=6.5999 \times 1.902=12.5530 (12-6/4')$   $R_4=6.5999 \times 1.902=12.5530 (12-6/4')$   $R_4=6.5999 \times 1.902=12.5530 (12-6/4')$   $R_4=6.5999 \times 1.902=12.5530 (12-6/4')$   $R_4=6.5999 \times 1.902=12.5530 (12-6/4')$ 

# ENGINEERING CALCULATION Poga No. 17 Noma C.Y.KING Data 412/198

	Small	Loadi	ng	
WKs	HE	W 16 TA		W 1/3
0.9		3.0951		1.457
0.8	•	2.75/2		1.4012
0.7		2,4073		1.3462
0.6		2,0634		1.2731
0.5		1.7195		1.1980
0.4		1.3756		1.1122
0,3		1.6317		1.0105
0.25		0,8598		0.9509
0,2		0.6878		0,8827
0.1		0.3439		0.7006
0:05	Ĺ	0,1720		0,5561
0.025		2.0860		0.4914

			***************************************
ENGINEERING	CALCULATION		Poga No.
		Noma	C.Y.KING
		Data	4/2//98

6. At I above the floor, to find the required distances from 2 inside of the anvil plate to the inside Surface of the wall. Page 11 at 1' W/3 = 1.4773 W=3.2241 = 19375 Kg

North wall (Left Portion) R=8,5' W3=1,4773 Z= E,5 7.0772 =5.7537 W=0.9 Kg R.9 = 5.7537 x 1.4573 = 8.3849 (8-45/8) R.8 = 5.7537 x 1.4012 = 8.0621' (8'-0)4") K.7 = 5.7537 × 1,3402 = 7.7111 (7-858) R.6 = 5.7537x 1.2731 = 7.3250 (7-4") R. 5 = 5.7537x 1.1980 = 6.8929' (6'-1034") F.4 = 5.7537x 1.1120 = 63981' (6'-4/8) R.3 = 5.7537 x 1.0105 = 11 (5'- 97/2) K.25 = 5.7537 x 0.9509 = 5.4712' (5'-5%) Riz = 5.7537 x0.8888 = 5.0788 (5'-1') R.1 = 5.7537x0,746 = 4.0310 (4'-0/x) Ries = 5.7537x 2000 = 3,1996 (3'-2/2) K, OZS = 5.7537 K 0, Jan = 2.5397' (2'-6/3)

North Wall (Right which)

R=19.5' W/3 = 1.4773 = 13.1998 W= c. 9 Kg = 13. 1888 x 14573 = 19.2361 (19-278") R.8=13.1998 C Adolz = 18.4956 (18'- 6") R.7 = 13,1943 + 13402 = 17,6904 (17-83")

## ENGINEERING CALCULATION Roge No. 19 Nome C.Y.KING Dote 4/21/98

 $R.6 = 13.1998 \times 1.2731 = 16.8047 \quad (16'-9'/4'')$   $R.5 = 13.1998 \times 1.1980 = 15.8134 \quad (15'-9'/6'')$   $R.4 = 13.1998 \times 1.1120 = 14.6782 \quad (14'-8/4'')$   $R.3 = 13.1998 \times 1.0105 = 13.3384 \quad (13'-4/8')$   $R.25 = 13.1998 \times .9509 = 12.5517 \quad (12'-6'/8')$   $R.2 = 13.1998 \times .8827 = 11.6515 \quad (11'-7'/8'')$   $R.1 = 13.1998 \times .7006 = 9.2478 \quad (9'-3'')$   $R.05 = 13.1998 \times .5561 = 7.3404 \quad (7'-4/8')$   $R.025 = 13.1998 \times .4414 = 5.8264 \quad (5'-10'')$ 

 $\frac{W(s+W_{\alpha})!}{R=9.37'} = \frac{W(s+W_{\alpha})!}{W(s)^{2}=1.4773} = \frac{9.37}{1.4773} = 6.3427$   $R.9 = 6.3427 \times 1.4573 = 9.2432 \quad (9'-3'')$   $R.8 = 6.3427 \times 1.4012 = 8.8874 \quad (8'-10\frac{3}{4}')$   $R.7 = 6.3427 \times 1.3402 = 8.5005 \quad (8'-6\frac{6}{6}'')$   $R.6 = 6.3427 \times 1.2751 = 8.0749 \quad (6'-1'')$   $R.5 = 6.3427 \times 1.1980 = 7.5986 \quad (7'-7\frac{4}{4}')$   $R.4 = 6.3427 \times 1.1120 = 7.0531 \quad (7'-0\frac{3}{4}'')$   $R.3 = 6.3427 \times 1.0105 = 6.4693 \quad (6'-5'')$   $R.15 = 6.3427 \times 1.9509 = 6.6313 \quad (6'-0\frac{3}{6}')$   $R.1 = 6.3427 \times .8827 = 5.5981 \quad (5'-7\frac{4}{4}'')$   $R.1 = 6.3427 \times .7006 = 4.4437 \quad (4'-5\frac{3}{6}'')$   $R.05 = 6.3427 \times .7006 = 4.4437 \quad (4'-5\frac{3}{6}'')$   $R.05 = 6.3427 \times .7006 = 3.5272 \quad (3'-6\frac{3}{6}'')$   $R.05 = 6.3427 \times .7006 = 3.5272 \quad (3'-6\frac{3}{6}'')$ 

## ENGINEERING CALCULATION Poge No. 20 Nome C.Y.KING Dote 4/21/98

South wall R = 19.5'  $W^3 = 1.4773$   $Z = \frac{19.5}{1.4773} = 13.1998$   $R.9 = 13.1998 \times 1.4573 = 19.2361 (19.2.2.76")$   $R.8 = 13.1998 \times 1.4012 = 18.4956 (18.6")$   $R.9 = 13.1998 \times 1.3902 = 17.6969 (17.6.3.6")$   $R.6 = 13.1998 \times 1.2731 = 16.8047 (16.9.3.6")$   $R.5 = 13.1998 \times 1.1980 = 15.8134 (15.9.7.6")$   $R.6 = 13.1998 \times 1.1120 = 14.6782 (14.6.6.6")$   $R.7 = 13.1998 \times 1.1120 = 14.6782 (14.6.6.6)$   $R.7 = 13.1998 \times 1.1120 = 14.6782 (14.6.6.6)$   $R.7 = 13.1998 \times 1.1120 = 14.6782 (14.6.6.6)$   $R.7 = 13.1998 \times 1.1120 = 14.6782 (11.6.6.6)$   $R.7 = 13.1998 \times 1.1120 = 12.5117 (12.6.6.6)$   $R.7 = 13.1998 \times 1.9509 = 12.5117 (12.6.6.6)$   $R.7 = 13.1998 \times 1.9509 = 12.5117 (12.6.6.6)$   $R.7 = 13.1998 \times 1.9509 = 12.5117 (12.6.6.6)$   $R.7 = 13.1998 \times 1.9506 = 17.3404 (7.6.6.6)$   $R.7 = 13.1998 \times 1.9506 = 17.3404 (7.6.6.6)$   $R.7 = 13.1998 \times 1.9506 = 17.3404 (7.6.6.6)$ 

East wall  $R=13' w's = 1.4773 = \frac{13}{8.7998} = 8.7998$   $Rig = 8.7998 \times 1.4573 = 12.8239' (126-10'')$   $Rig = 8.7998 \times 1.4012 = 12.3303' (126-4'')$   $Rig = 8.7998 \times 1.3402 = 11.7938 (126-4'')$   $Rig = 8.7998 \times 1.2731 = 11.2010 (126-25'')$   $Rig = 8.7998 \times 1.1980 = 108020 (126-65'')$   $Rig = 8.7998 \times 1.1120 = 9.7880 (126-912'')$   $Rig = 8.7998 \times 1.0105 = 0.8820 (136-1034'')$   $Rig = 8.7998 \times 1.0105 = 0.8820 (136-1034'')$   $Rig = 8.7998 \times 1.0105 = 0.8820 (136-1034'')$ 

ENGINEERING	CALCULATION		Poga No.	
		Nome	C.Y.KING	
		Data	4/21/90	

$$R_{12} = 8.7998 \times .8827 = 7.7676 (7'-9/4'')$$

$$R_{11} = 8.7998 \times .7006 = 6.1651 (6'-2'')$$

$$R_{10} = 8.7998 \times .5561 = 4.8936 (4'-103/4'')$$

$$R_{0.15} = 8.7998 \times .4414 = 3.8842 (3'-103/8'')$$

### 7 To determine stand-off Distance.

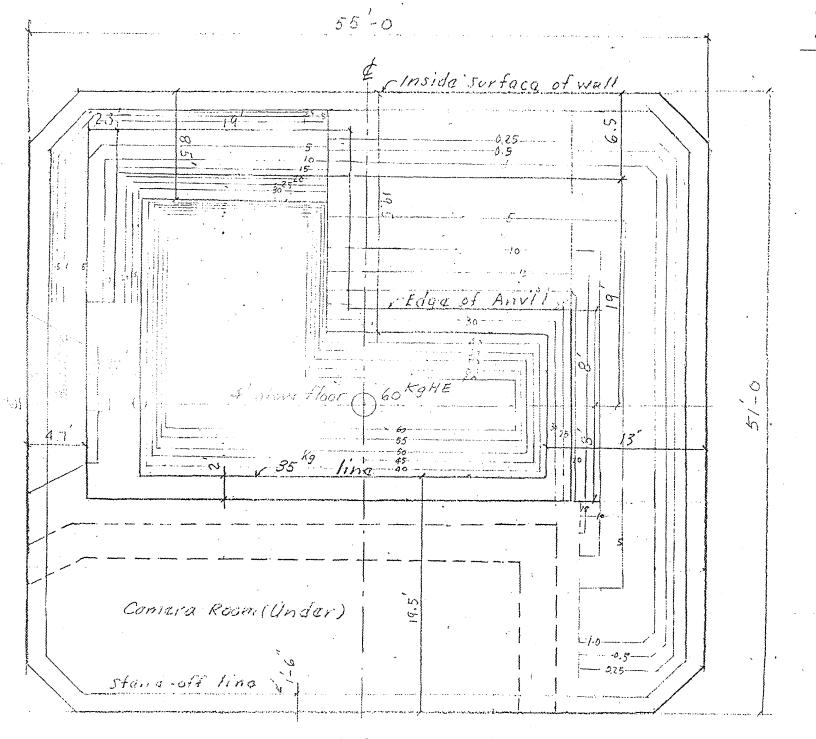
8. To find the distance from the floor where the stress of the cailing is 4 ksi

Concrete compression stress of floor = 12 ksi

" Cailing = 4 ksi

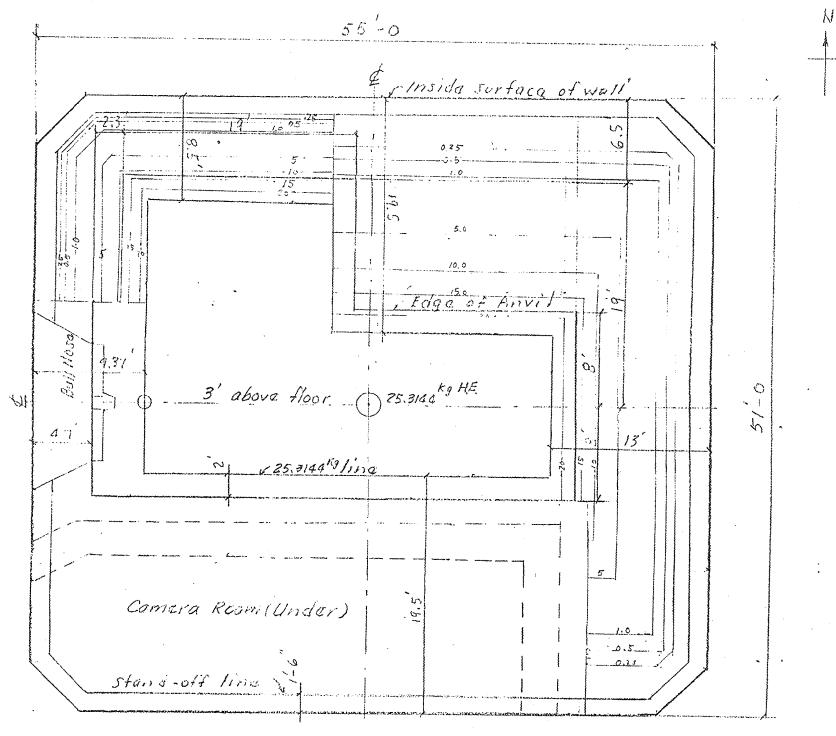
 $\frac{2}{2} floor = \frac{4}{5.9c9} = .677$   $\frac{12}{2} coiling = .677 \times \frac{12}{4} = 2.031$   $\frac{R}{5.9c9} = 2.031 \quad R = 2.031 \times 5.9c9 = 12.03 \quad (from Cailing)$   $30' - 12.03' = 17.97 \quad (from floor)$ The electrotal Curves from 4' up to 17.97' are
the same as 4'.

9. Above Camara voom, limit W=150, to prevent damage of the Camera lens.



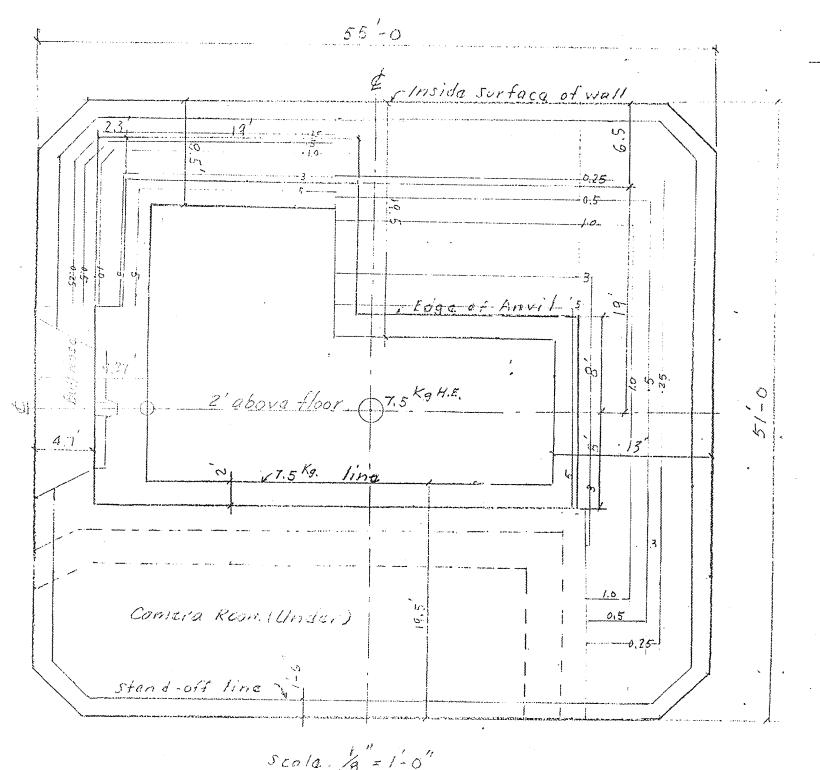
1 6

Scala Kelio

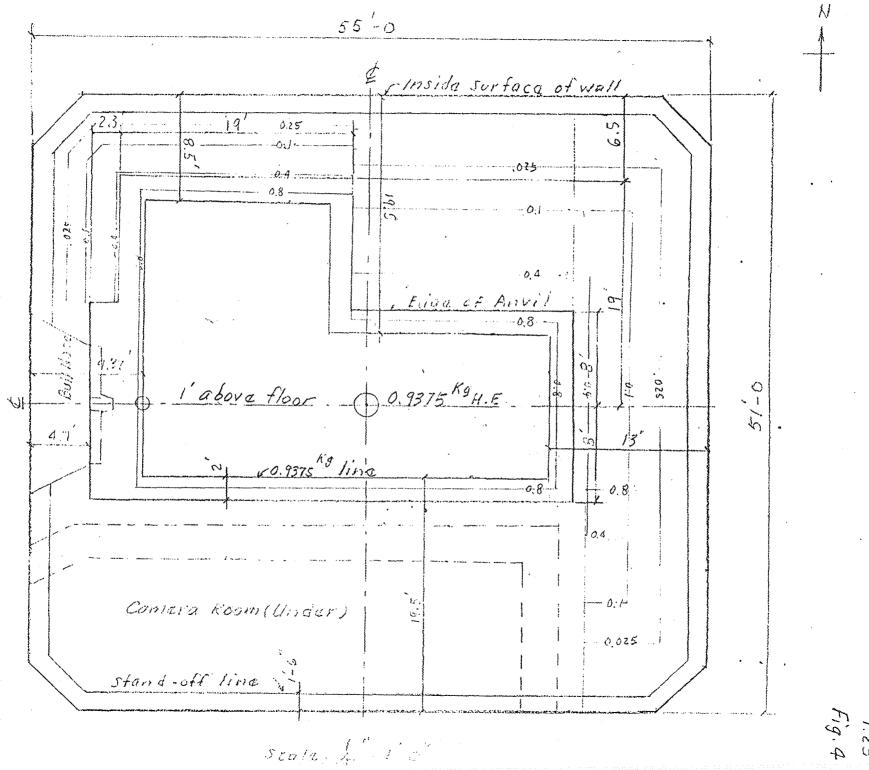


F19.2

Scala 12 1:0"



P. 24 Eig 3



ENGINEERING	CALCULATION		Poga No.
		Noma	. 26
		Data	

#### Raferences

- 1. Parsons Infrastructura & Tachnology Group "CFF" Title I, 100 % Design Review.
- 2. TM 5-1300, NAVFAC P-397, AFB 88-22" Structures to resist the effects of Accidental Explosions."